





The NET Power Cycle and the Combustor and Turbine Development

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Toshiba Corporation
Power Systems Company

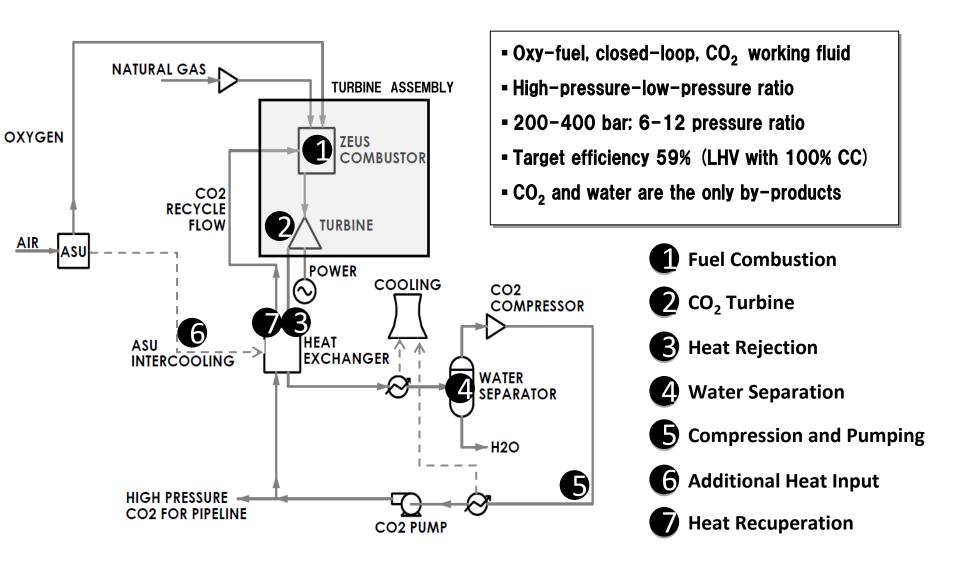
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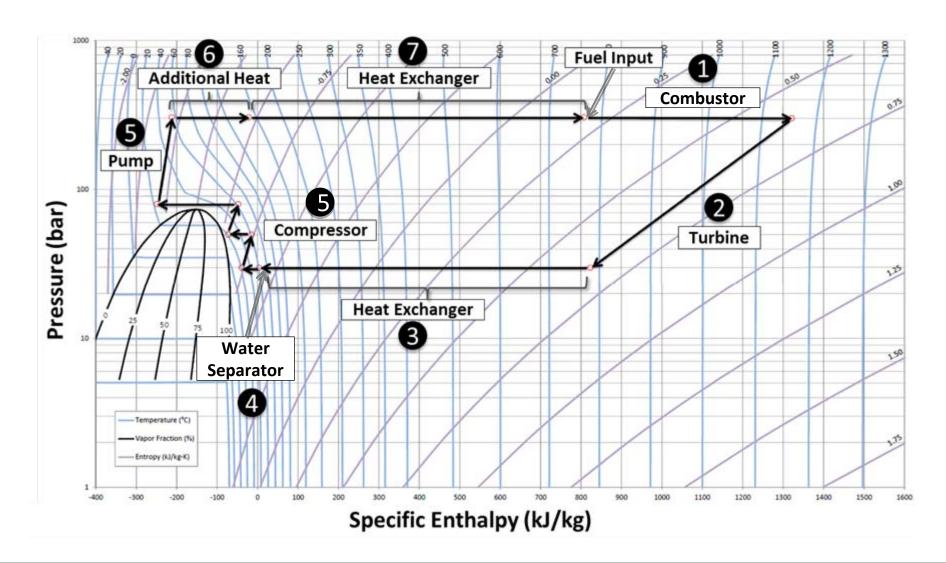
Natural Gas Cycle: The Platform







Pressure and Enthalpy Diagram







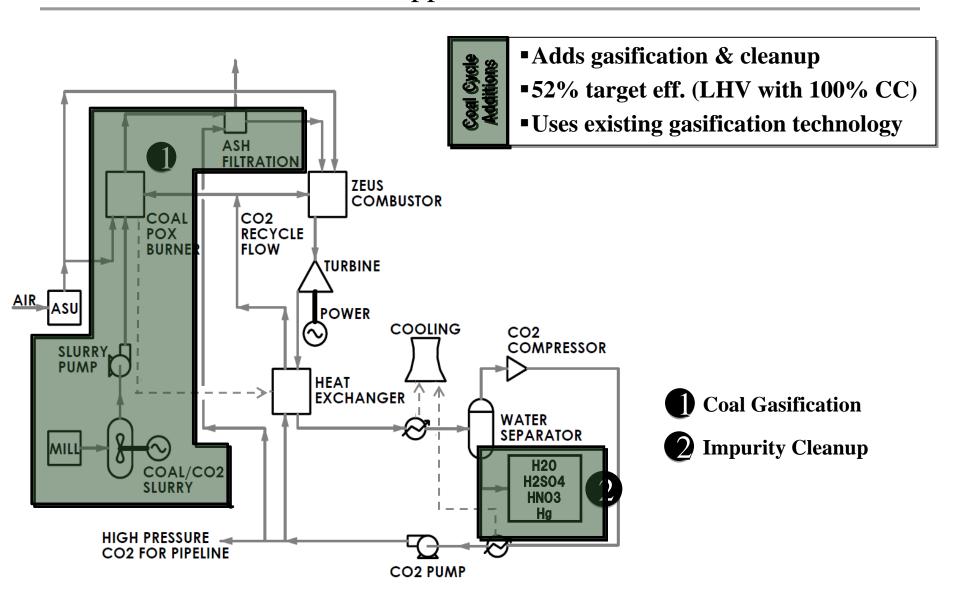
NET Power Platform Target Efficiency

Natural Gas Platform Target Efficiencies (100% CO ₂ Capture at 300 bar)		
Energy Components	нну	LHV
Gross Turbine Output	75%	83%
CO ₂ Compressor Power	-11%	-12%
Plant Parasitic Power (primarily ASU)	-11%	-12%
Net Efficiency	53%	59%





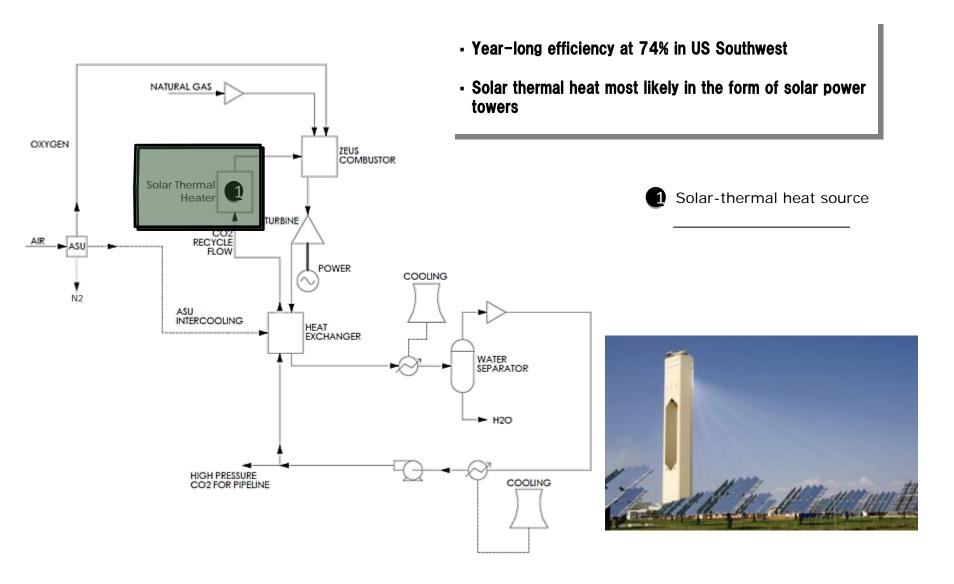
Coal Application





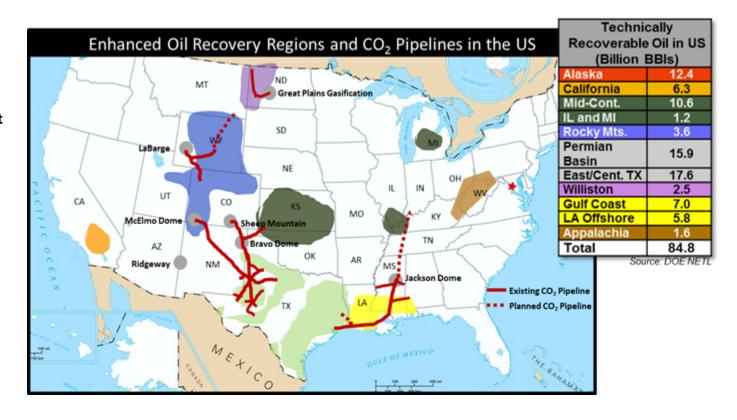


Direct Solar Integration



EOR Market: A Large Financial and CO₂ Storage Opportunity

85 billion barrels technically recoverable in the US; industry is tethered to current pipeline and geologic CO₂ infrastructure



470-1,000 billion barrels of oil technically recoverable globally

GLOBAL CO ₂ EOR POTENTIAL		
Region	Billion Bbls	
Middle East	230	
Russia	78	
United States	85	
S. America	32	
Asia Pacific	18	
Europe	16	
Africa	15	



Assuming a plant size of 550 MW, this need would support the CO₂ production from 1382 NET Power gas plants (691 coal).





Four Way Agreement and Commercial Relationships

NET Power*

Inventor and developer of the technology. Responsible for overall project development, systems engineering and commercial development

Toshiba*

Developing the turbine and combustor

Goodwin Steel*

CB&I (Shaw*)

Provided substantial investment in this project and performing EPC services.

Exelon

Assisting with the siting, permitting, and commissioning of the natural gas demo facility; providing operations and maintenance support.

*Note; UK grant was awarded for the development.

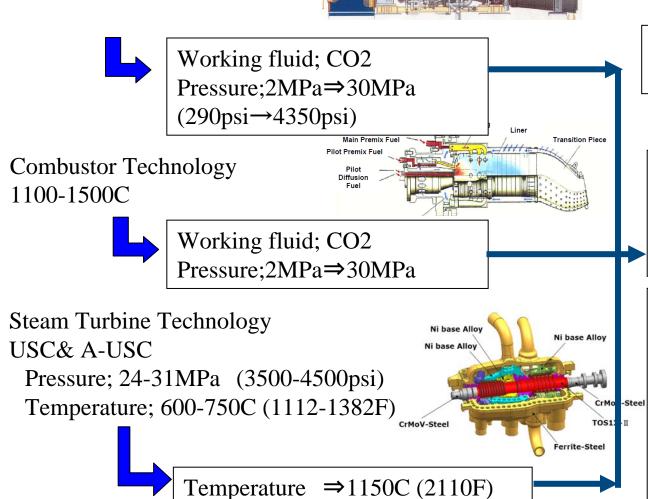




Necessary Technology for NET-Power Turbine

Gas Turbine Technology 1100-1500C (2012-2730F)





Temperature; E-Class Pressure; USC & A-USC

Turbine & Combustor for Net Power

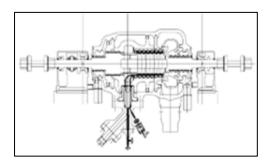
Temp. 1150C (2110F)

Press. 30MPa (4350psi)

Toshiba is the only company that manufactured commercial turbine of 31Mpa main steam pressure.
Toshiba has been keen on A-USC development.

Turbine for 25MWe Demo Plant

- A) Intended to be a scale model of commercial turbine (250MWe) as much as possible
- B) Rotational speed is 6000rpm and connected to compressor and reduction gear
- C) Double shell configuration
- D) Rotors are welded together
- E) Single can type combustor for 25MWe turbine



Materials

Rotor; Ni base forging and CrMoV forging are welded together

Casings; Ni base casting for high temperature part

CrMoV casting for lower temperature part

Blades; Ni base casting

All the necessary materials have been already developed. Purchase orders for long-lead materials will be placed soon.



R&D results of Ni base Forging and Ni base Casting - Make the best use of R&D results for A-USC -



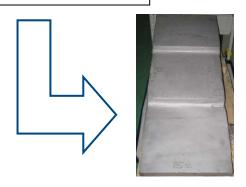


First trial forging for TOS1X was completed (above photograph) Second trial forging for rotational test will be manufactured soon.



Alloy625
Trial Inner casing for A-USC



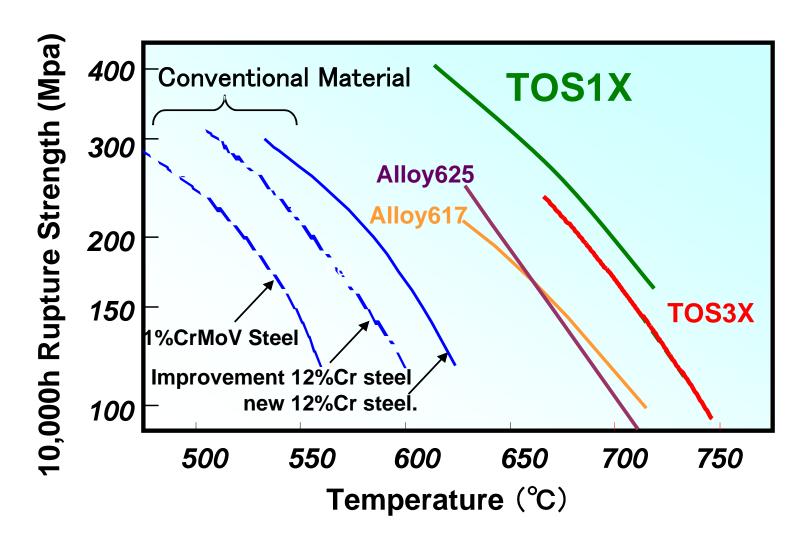


TOS3X Test Piece

Two candidates, Alloy 625 and TOS3X, are available for S-CO2 Turbine



Comparison of Creep Rupture Strength



Necessary materials are already applicable



Cooling Design

High Temperature

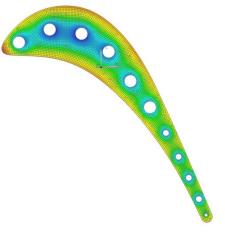


- ➤ Needs cooling both for nozzles and moving blades
- ➤ However, very complicated cooling technology is not necessary because the temperature is not extremely high compared with cutting-edge gas turbines.



Both mean temperature and local temperature satisfy design criteria thanks to two contributor

- Convection cooling by cold CO2
- Thermal barrier coating



Temp. Contour

Concept of Combustor Design

No NOx Emission

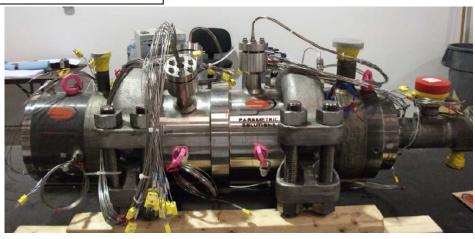
No need of complex pre-mix technology
Simple Diffusion Flame can be used

High Pressure

Thick Wall Casing against high Pressure

Rather moderate temperature compared with gas turbines





5MWt Rig Test Combustor

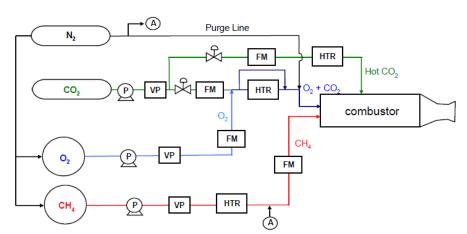
Rather simple combustion and cooling are expected, yet verification under high pressure using a rig test combustor is necessary.



Present Status of Combustor Development

- ✓ First Ignition was successfully done at the middle of January using test facility in U.S.A.
- ✓ Phase 1 Test up to 5Mpa has been completed.
- ✓ All the test data was carefully checked and evaluated.
- ✓ Stable flame was confirmed enabling us to proceed to higher pressure test (Phase 2).
- ✓ Modification of facility is being done for Phase 2 test.
- ✓ Design of the combustor for 25MW Demo plant will be synchronized with Turbine Development.

Propellant Blow-Down Configuration For Zeus Combustor testing





Rig Test Combustor at the Test Stand

END